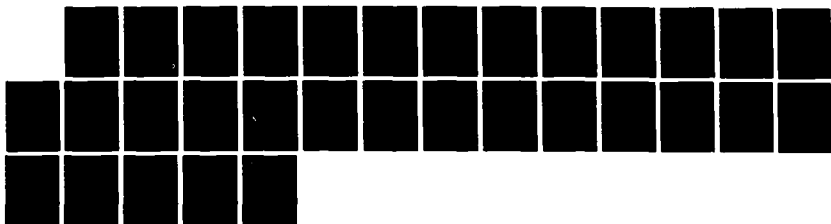
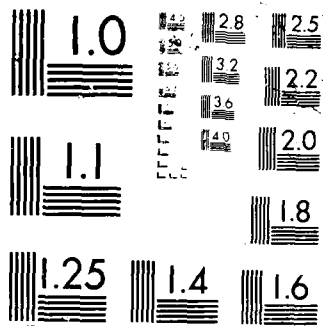


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**A Severe Spacecraft-Charging Event  
on SCATHA in September 1982**

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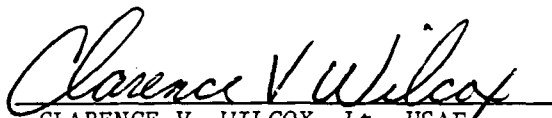
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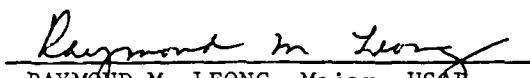
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This report has been reviewed by the Public Affairs Office (PAS) and is releasable to the National Technical Information Service (NTIS). At NTIS, it will be available to the general public, including foreign nationals.

This technical report has been reviewed and is approved for publication. Publication of this report does not constitute Air Force approval of the report's findings or conclusions. It is published only for the exchange and stimulation of ideas.

  
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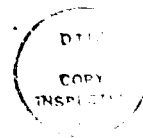
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<p>On September 22, 1982, 29 large amplitude electrostatic discharges were detected by the Pulse Analyzer onboard the SCATHA satellite. Seventeen of these pulses exceeded the maximum voltage discrimination level, which was set to 7.4 volts. This was the worst instance of electrostatic discharges encountered to date by the SCATHA satellite. Three different spacecraft anomalies occurred on SCATHA that day, the most serious being a two-minute loss of data. During this same time period, the Surface Potential Monitor experiment aboard the satellite measured the largest differential surface charging observed in the data since the satellite's launch in January 1979.</p>				
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## 1. INTRODUCTION

The P78-2 (SCATHA) satellite was launched on January 30, 1979. The primary objective of the satellite was to obtain environmental and engineering data that could be used to provide guidelines and material, process, and test specifications to ensure that future spacecraft will operate satisfactorily in a spacecraft charging environment. The experiments on the satellite are described by Stevens and Vampola<sup>1</sup> and Fennell.<sup>2</sup>

## II. PULSE ANALYZER DATA

A large number of intense electrostatic discharges were detected by the engineering instruments aboard the SCATHA satellite on 22 September 1982. The satellite was located at dawn very near the geosynchronous altitude. The Pulse Analyzer detected 29 pulses on that date. A summary of the pulses is given in Table 1. Seventeen of the pulses exceeded the maximum voltage discrimination level, which was set to 7.4 volts. The Transient Pulse Monitor also detected 29 pulses, as shown in Table 1. Most of these were coincident in time with those detected by the Pulse Analyzer. This was the worst instance of electrostatic discharges encountered to date by the SCATHA satellite. During this same time period, the Satellite Surface Potential Monitor (SSPM) experiment measured the largest differential surface charging observed in the data analyzed since launch.

The Pulse Analyzer has been operating continuously since it was turned on in February 1979. Since that time, data from 822 days (about 1/3 of the total) have been analyzed.<sup>3,4,5</sup> A total of 147 pulses attributed to electrostatic discharges on the vehicle have been detected. The 29 discharges detected on 22 September 1982 represent almost 20 percent of the total number of pulses detected in over two and one-half years of analyzed data.

The amplitude distribution of the discharge pulses is shown in Fig. 1. The voltage plotted along the x axis is the highest discriminator voltage level exceeded by each pulse. The discriminator levels are spaced by factors of approximately 2. The pulse amplitude distribution is unusual in that it shows a large number of pulses at the high voltage end of the distribution. Almost all of the pulses above one volt occurred on only one day, 22 September 1982. The amplitude distribution for 22 September 1982 is shown together with the total distribution in Fig. 1.

In Fig. 2 the time of the discharges (plotted as circles near the top of the figure) is compared with the absolute value of the differential potential of the gold sample on the SSPM with respect to the spacecraft ground. The discharges occurred when the potential of the gold sample was greater than

Table 1. Summary of discharges on 22 September 1982.

<u>UT,s</u>	<u>Pulse Analyzer</u>			<u>TPM</u>		
	<u>LT</u>	<u>Re</u>	<u>Sensor*</u>	<u>Volts</u>	<u>+Amp</u>	<u>-Amp</u>
5109	22.7	7.3	3	0.12		
24059	3.8	5.8	3	7.4	2.82	3.40
24118	3.8	5.8	3	7.4	4.27	5.35
24177	3.8	5.8	3	7.4	5.16	5.56
24235	3.8	5.8			5.77	6.47
24238	3.8	5.8	3	7.4	5.16	6.00
24297	3.8	5.8	1	1.9	6.47	7.52
24356	4.0	5.8	1	1.9	5.16	5.56
24360	4.0	5.8	1	1.9	3.96	5.77
24419	4.0	5.8	1	3.8	6.97	7.18
24479	4.0	5.8	1	3.8	6.47	6.97
24539	4.0	5.8	1	1.9	6.23	6.71
24596	4.0	5.8	0	7.4	4.97	5.35
24600	4.0	5.8	0	7.4	6.47	7.52
24658	4.0	5.8	0	7.4	6.47	8.42
24716	4.0	5.8	0	7.4	5.56	8.42
24778	4.0	5.8	2	7.4	3.67	4.11
24837	4.0	5.8	2	7.4	2.61	3.28
24900	4.2	5.7	2	7.4	3.40	3.28
25381	4.2	5.7	1	1.9	5.56	5.77
25498	4.2	5.7	0	7.4	5.35	6.23
25558	4.4	5.7	0	7.4	4.97	5.16
25616	4.4	5.7	0	7.4	5.77	6.47
25620	4.4	5.7	0	7.4	4.27	5.35
25678	4.4	5.7	0	7.4	5.77	6.71
25738	4.4	5.7	2	7.4	3.40	4.78
26000	4.4	5.7			6.00	6.71
27238	4.9	5.6	1	1.9	7.24	7.52
27244	4.9	5.6	3	0.7		
27300	5.1	5.6	1	3.8	6.23	7.81
27359	5.1	5.6	1	1.9	6.23	7.81

Sensors:

- 0 external dipole
- 1 loop around the Command Distribution Unit
- 2 harness wire
- 3 command line

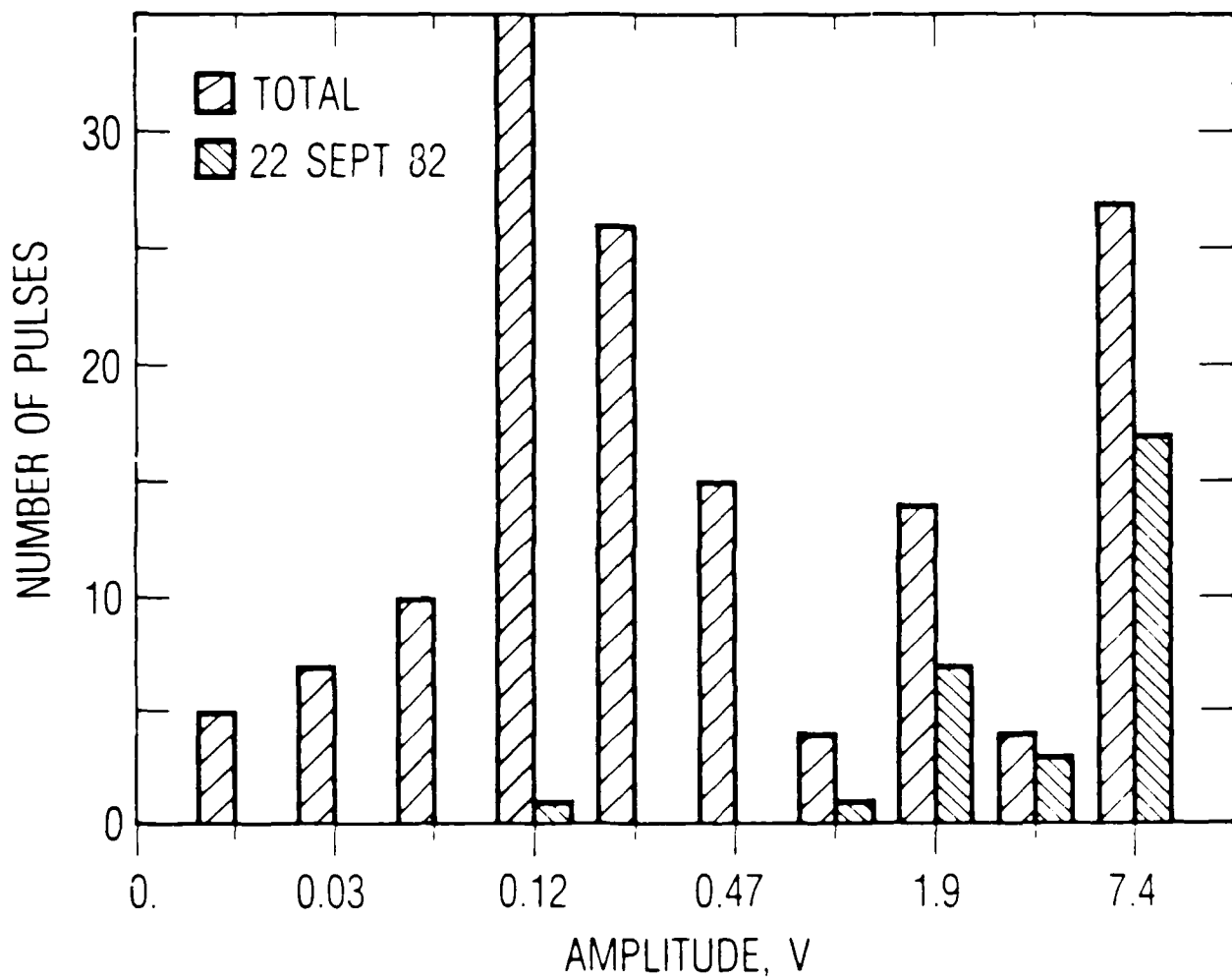


Fig. 1. Amplitude Distribution of Pulses Caused by Electrostatic Discharges, with Pulses Detected on 22 September 1982 Shown Separately

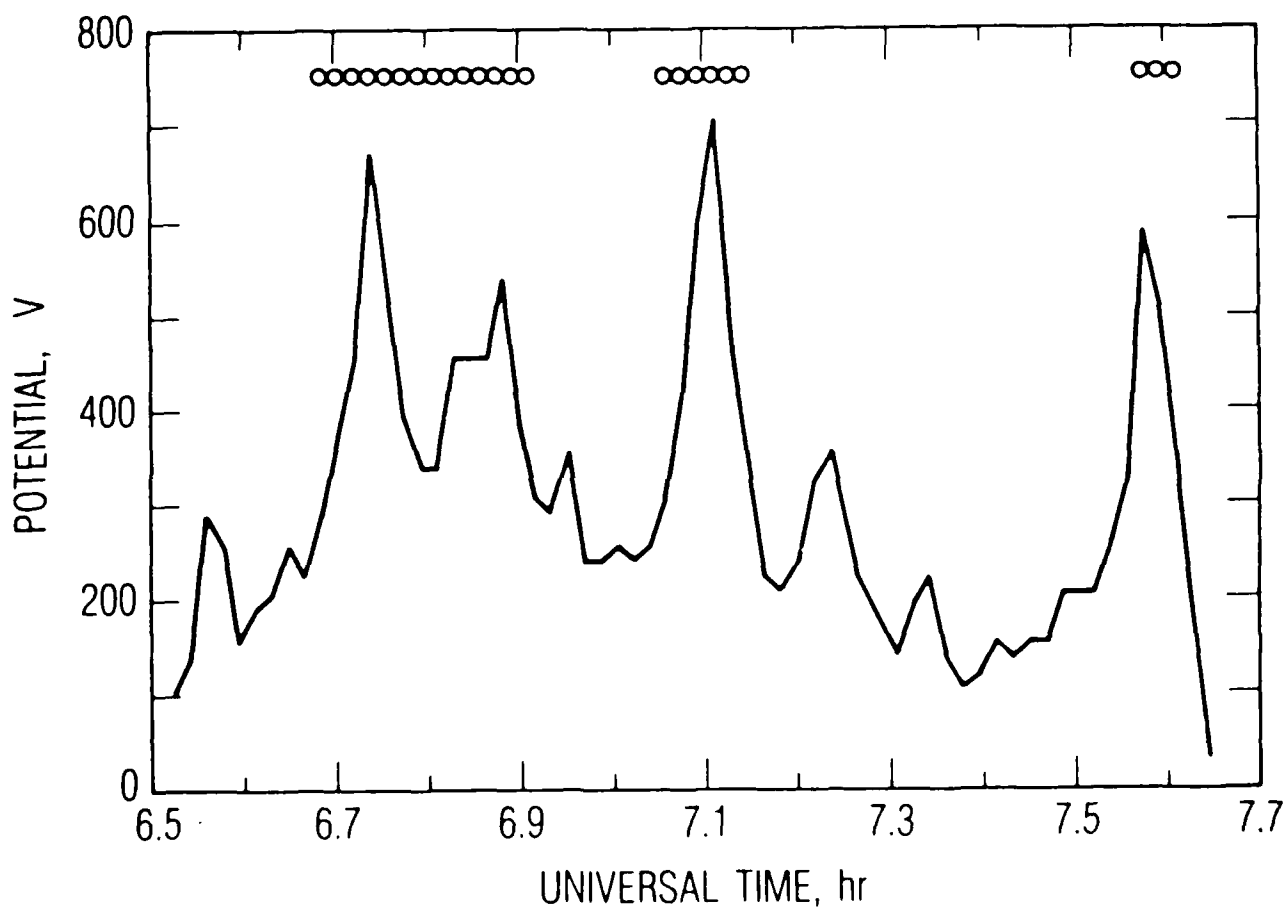


Fig. 2. Absolute Value of the Potential of the Gold Sample on the Satellite Surface Potential Monitor with Respect to the Space Vehicle Ground Reference Point. The Potential of the Sample is Negative with Respect to the Ground Reference. The Circles Plotted Near the Top of the Figure Identify the Times When the Pulse Analyzer Detected Discharges.

-300 volts with respect to the reference ground. Although there is no evidence that this particular sample was responsible for the discharges, this is the most direct evidence available that discharges on SCATHA were due to differential potentials on the vehicle.

In Fig. 3 the amplitude distribution for the 22 September discharges is compared with the amplitude distribution for the factory test pulses measured during the systems level electrostatic discharge tests. The system level tests were conducted in accordance with MIL-STD-1541. Almost all of the pulses on 22 September exceeded the factory test pulses. We conclude that the factory test pulses are not large enough in amplitude to represent a worst-case on-orbit environment.

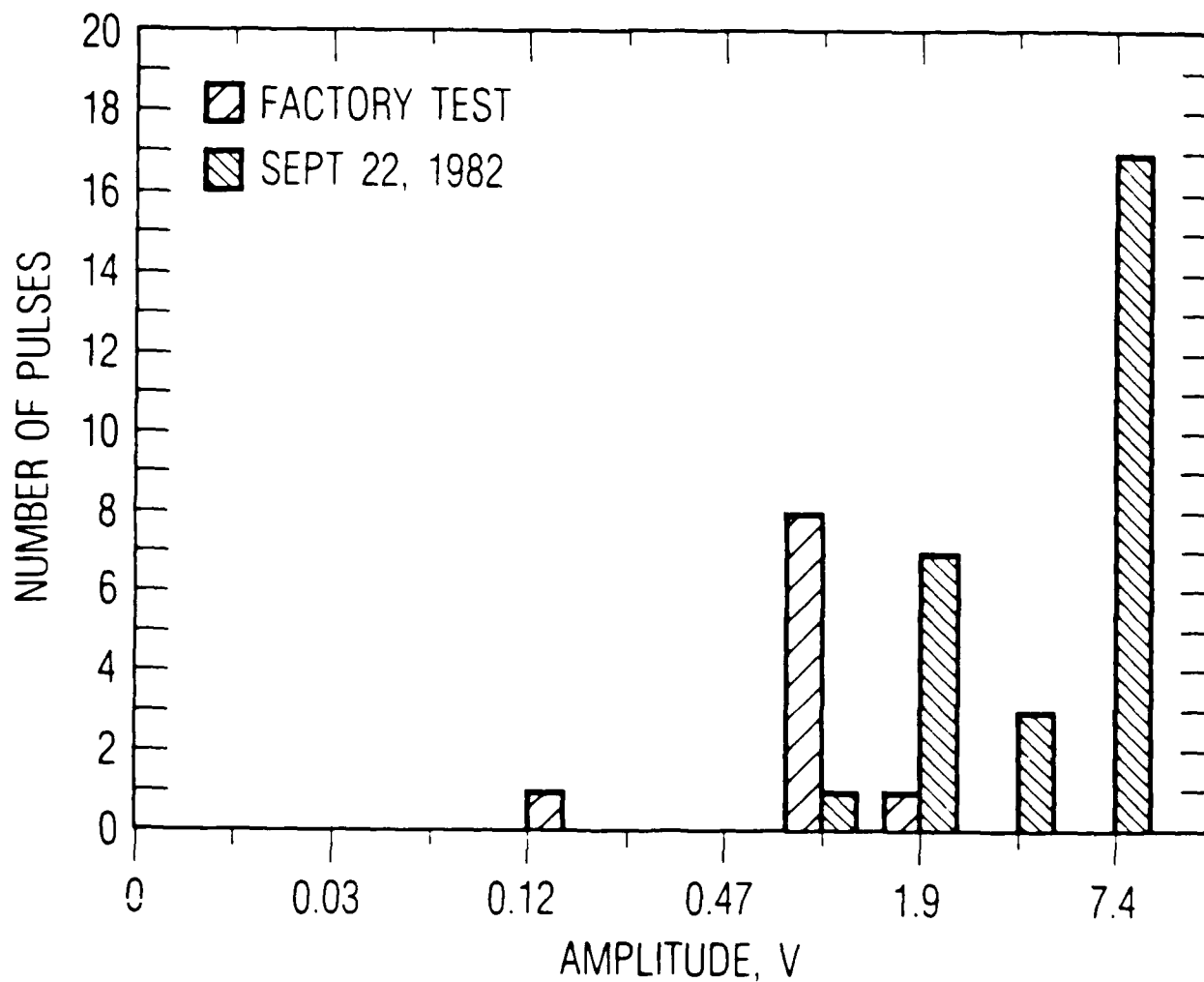


Fig. 3. Amplitude Distribution of the Pulses Detected on 22 September 1982 is Compared with the Amplitude of the Pulses Measured During the System Level Factory Electrostatic Discharge Tests



### III. SURFACE POTENTIAL MONITOR DATA

Figure 4a shows the time plot of the absolute value of the gold sample voltage beginning at 02 UT and ending at 08 UT on 22 September 1982. The voltage is negative with respect to the satellite ground. The gold sample experienced low-level differential charging between 02 and 03 UT. The SCATHA satellite entered the eclipse of the earth shortly after 02:30 UT. The differential potential of gold remained relatively constant before and during the eclipse at values between -100 to -200 volts. As the satellite proceeded into the post-midnight local time sector, the differential voltage exceeded -300 volts near 04:15 UT (note: local time = universal time minus 3 hr). The charging decreased over the next two hours and then again increased around 06:30 UT. The gold sample reached voltages greater than -300 volts on four separate occasions. Each time the voltage exceeded this value, electrostatic discharges were recorded by the Pulse Analyzer.

Figure 4b shows the logarithm of the 18.4 keV electron intensity recorded by the Energetic Ion Composition Experiment. There is qualitative agreement between increases in the electron flux and the differential charging levels.

Figure 5a shows the time history of the Teflon and the quartz fabric sample voltages plotted with the logarithm of the 140 keV electron intensity in Figure 5b. Electrons with energies  $\geq 60$  keV were measured by the High Energy Particle Spectrometer experiment. Electrons of these high energies do not provide the majority of the charging current to the samples, but they do play a role in modifying the charging properties of dielectrics such as Teflon.<sup>6</sup> Therefore, the magnitude of the differential charging may well depend on the high energy tail of the electron distribution.

The quartz fabric has a tendency to charge more frequently and to larger values than the Teflon sample. A differential potential of nearly 10,000 volts measured on the quartz sample is the largest value recorded by the SSPM instrument. The potentials of the Teflon and quartz fabric samples are the maximum values reached during one satellite rotation, which took 1 minute. The SSPM-3 instrument containing the Teflon and quartz fabric samples was

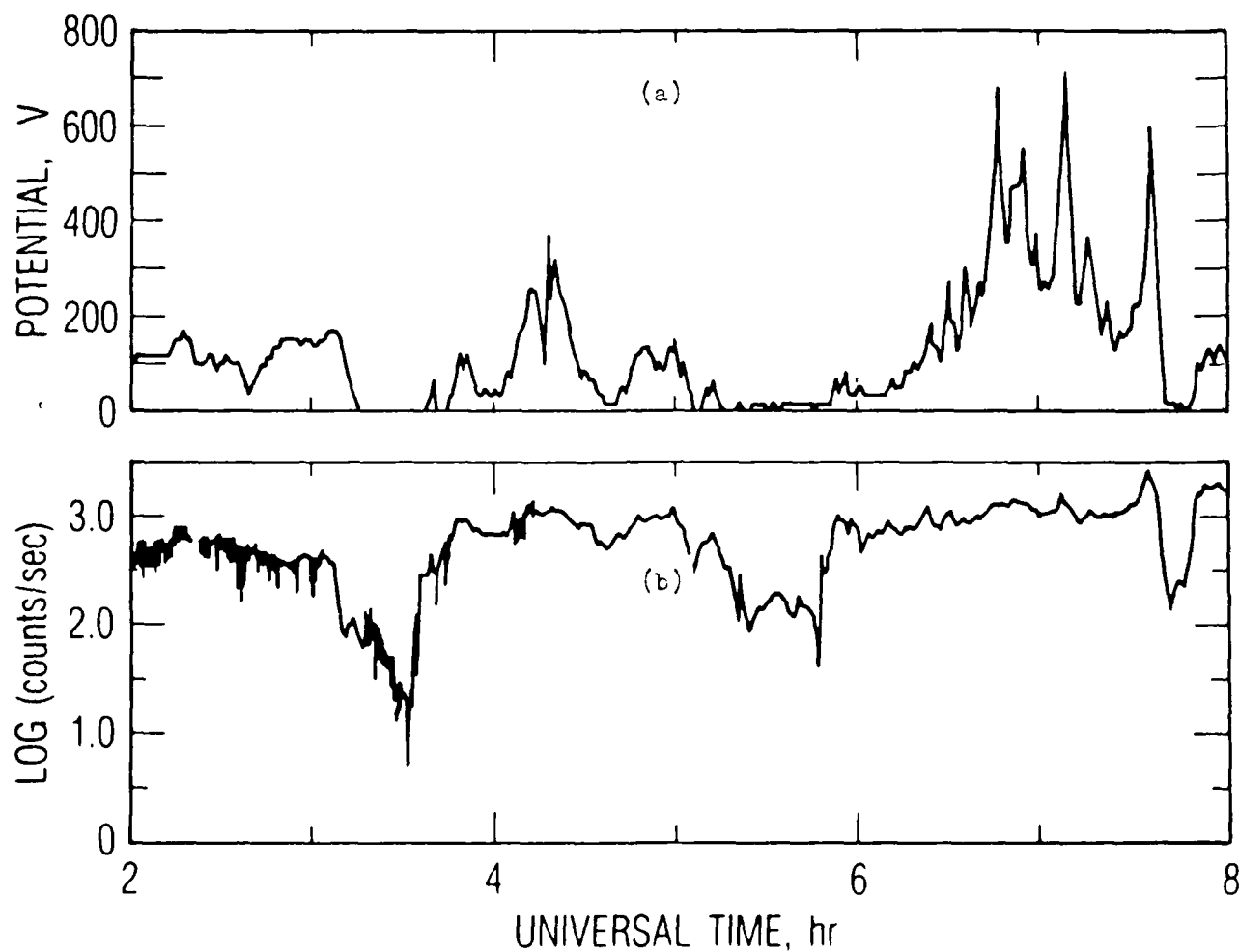


Fig. 4. (a) Absolute Value of the Voltage on the Satellite Surface Potential Monitor Gold Sample as a Function of Time. The Potential of the Sample is Negative with Respect to the Ground Reference. (b) the count rate for 18.4 keV electrons as a function of time

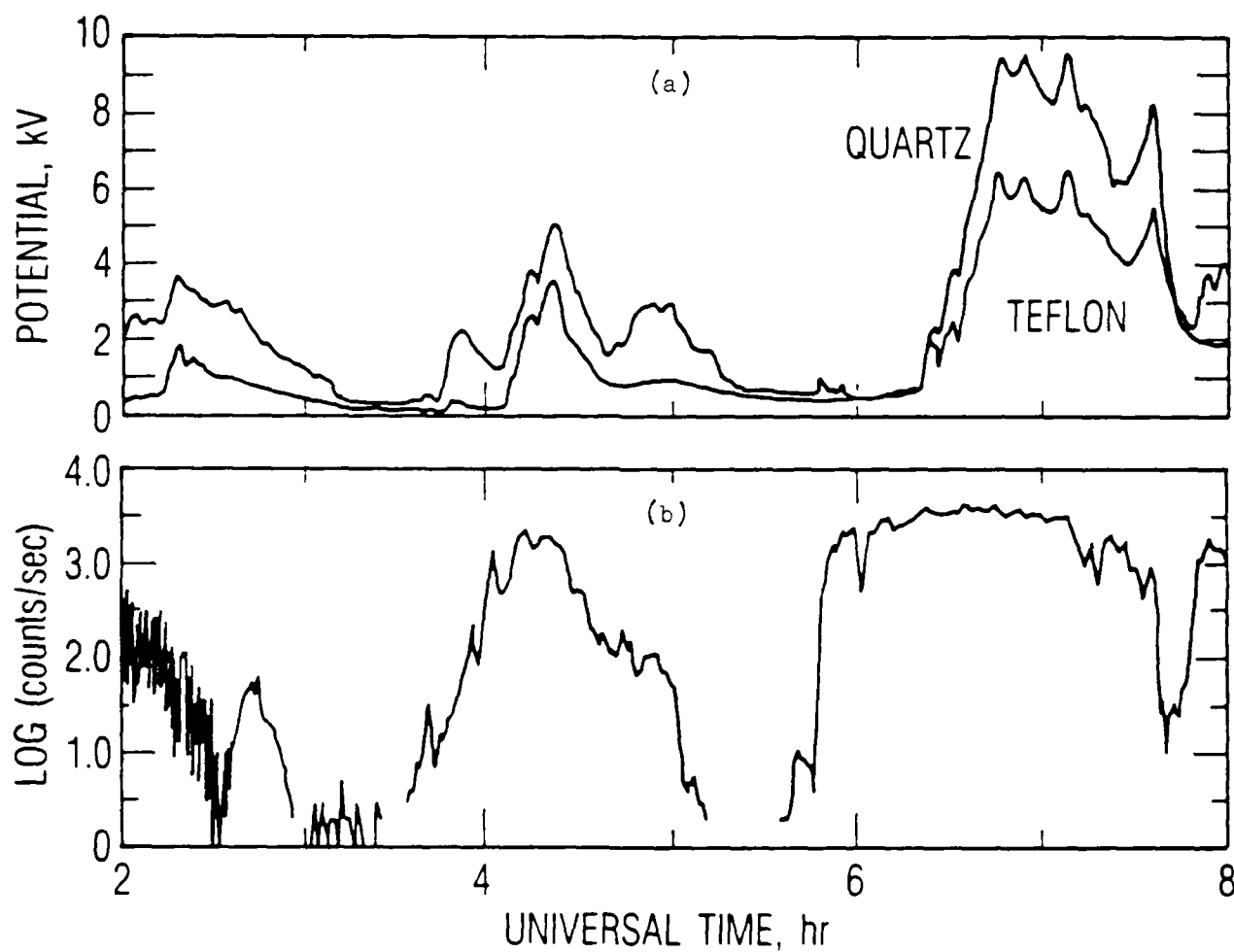


Fig. 5. (a) Absolute Value of the Voltage on the Satellite Surface Potential Monitor Quartz and Teflon Samples as a Function of Time. The Potential of the Sample is Negative with Respect to the Ground Reference. (b) the Count Rate for 140 keV Electrons as a Function of Time

positioned so that the samples pointed along the spin axis, which was maintained perpendicular to the satellite-sun direction to within 5 deg. In contrast, the gold sample on the SSPM-1 instrument rotated into and out of sunlight every 30 seconds, causing it to discharge before reaching its maximum value in shadow.

Figure 2 shows an expanded time plot of the gold sample from 6.5 to 7.7 hr UT. The circle symbols represent times when discharges were recorded by the Pulse Analyzer. The discharges occurred with time differences very near the satellite rotation period of 60 seconds. The data in Fig. 2 suggest that a threshold of -300 volts must be exceeded on the gold sample before discharges occurred somewhere on the satellite. There is a good correlation of the onset of the three charging periods (6.75, 7.12 and 7.58 hr UT) with the onset of the electrostatic discharge pulses.

Table 2 shows a list of selected sample voltages measured by the SSPM on 24 April 1979 and 22 September 1982. The earlier time was selected to represent the worst-case environment as reported by AFGL.<sup>7</sup> For all samples, except the Kapton sample on the SSPM-3 instrument on the end of the spacecraft, the 22 September 1982 voltages were larger than for the 24 April 1979 event. Since the conductivity of the Kapton samples on the SSPM-1 and -2 instruments on the side of the vehicle changed over a period of months due to solar exposure, we would expect a similar effect for the Kapton sample located on the end of the spacecraft to occur, but over a longer period of time. This increase in the conductivity of the Kapton may account for the lower voltages observed during the 1982 event.

Table 2. Differential Voltages

<u>Date</u>	<u>UT</u>	<u>Au</u>	<u>OSR</u>	<u>Kapton</u>	<u>Teflon</u>	<u>Quartz Fabric</u>
22 Sept 1982	06:44	-650	-150	-250	-6250	-9400
22 Sept 1982	07:06	-680	-150	-285	-6350	-9550
22 Sept 1982	07:34	-570	-210	-250	-5500	-8260
24 April 1979	06:52	-400	-150	-1550	-3400	-1700

#### IV. SPACECRAFT ANOMALIES

Three different spacecraft anomalies occurred on SCATHA on 22 September 1982. The most serious was a two-minute loss of data. The other two were uncommanded mode changes in two experiments.

Two minutes are missing from the tape-recorded data from the satellite from 26159 to 26271 s UT. Although no pulse was recorded near that time by the pulse detectors, the signature of a pulse appears in the data from the Very-Low-Frequency (VLF) Wave Analyzer experiment. Each pulse detected by the Pulse Analyzer also produced an anomalous output from the VLF experiment. The pulses appear in the VLF data as 99 dB, while the actual calibrated values are all less than 0 dB. At 26158.6 s, a 99 appears in the VLF data. Apparently a discharge occurred at approximately 26158 s. We believe that the discharge caused the PCM system to lose synchronization. The pulse shape as measured by the Pulse Analyzer would have been read out during the next second of data. Since these data could not be recovered, the pulse does not appear in the data from the Pulse Analyzer.

Two experiments on SCATHA experienced anomalies that coincided with discharges detected by both the Pulse Analyzer and the Transient Pulse Monitor.

An anomalous reconfiguration of the Magnetic Field Monitor experiment occurred, and anomalous timing errors occurred in the VLF Plasma Wave Analyzer.

A filter select relay in the Magnetic Field Monitor experiment occasionally changes state. This occurs during time periods when discharges are occurring on the vehicle. One such filter change occurred on 22 September.

The VLF experiment collects data from two sensors, an electric antenna and a magnetic antenna. The experiment contains a counter that counts the Main Frame sync pulses from the telemetry system. These occur at the rate of one per second. Every 16 sync pulses (seconds), the antenna is switched. The

anomaly that occurred on 22 September was a failure of the experiment to switch properly after virtually every discharge. The observed and expected switching sequences for one time period are shown in Fig. 6. The switching failures in the figure coincided with four of the discharge pulses between 27238 and 27359 s UT.

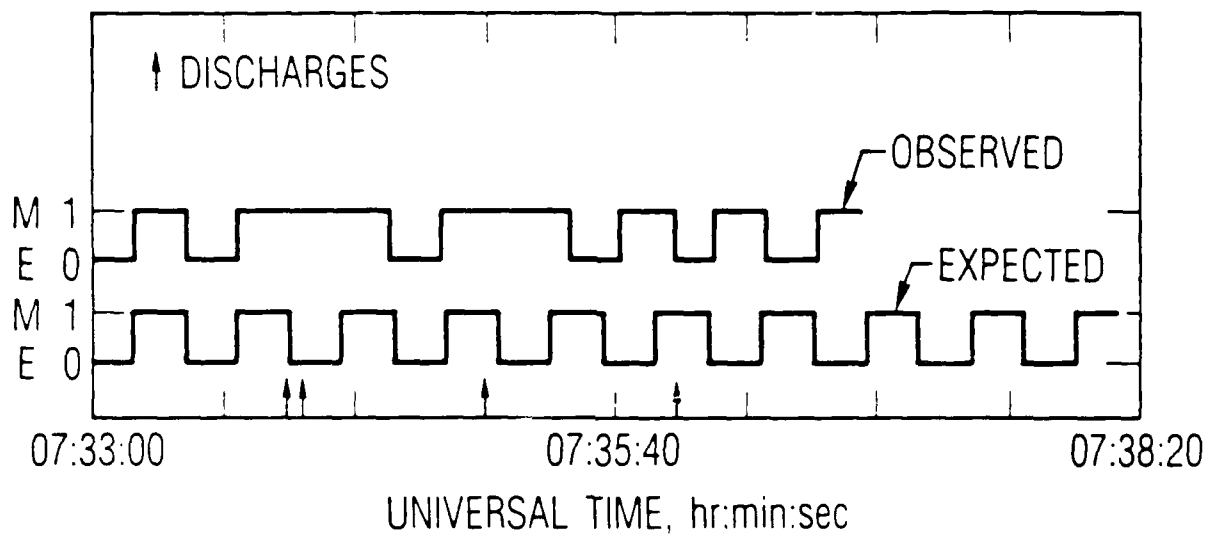


Fig. 6. Timing Sequence of the Antenna Switch on the VLF Analyzer Experiment During One Time Period When Anomalous Behavior Occurred on 22 September 1982

## V. ASPECT DEPENDENCE

The 29 discharges detected on 22 September between 24059 and 27359 s UT occurred at very nearly the same rotational phase of the vehicle. A histogram of the period between two consecutive discharges is shown in Fig. 7. The peak at 60 seconds coincides with the spin period of the satellite. Since the satellite was in sunlight (the local time was 04 hr), the locking of the discharges to the spin rate suggests that only one location on the vehicle was arcing. Similar phenomena occurred on 26 May 1979 when six discharges occurred with a periodicity equal to the spin period.



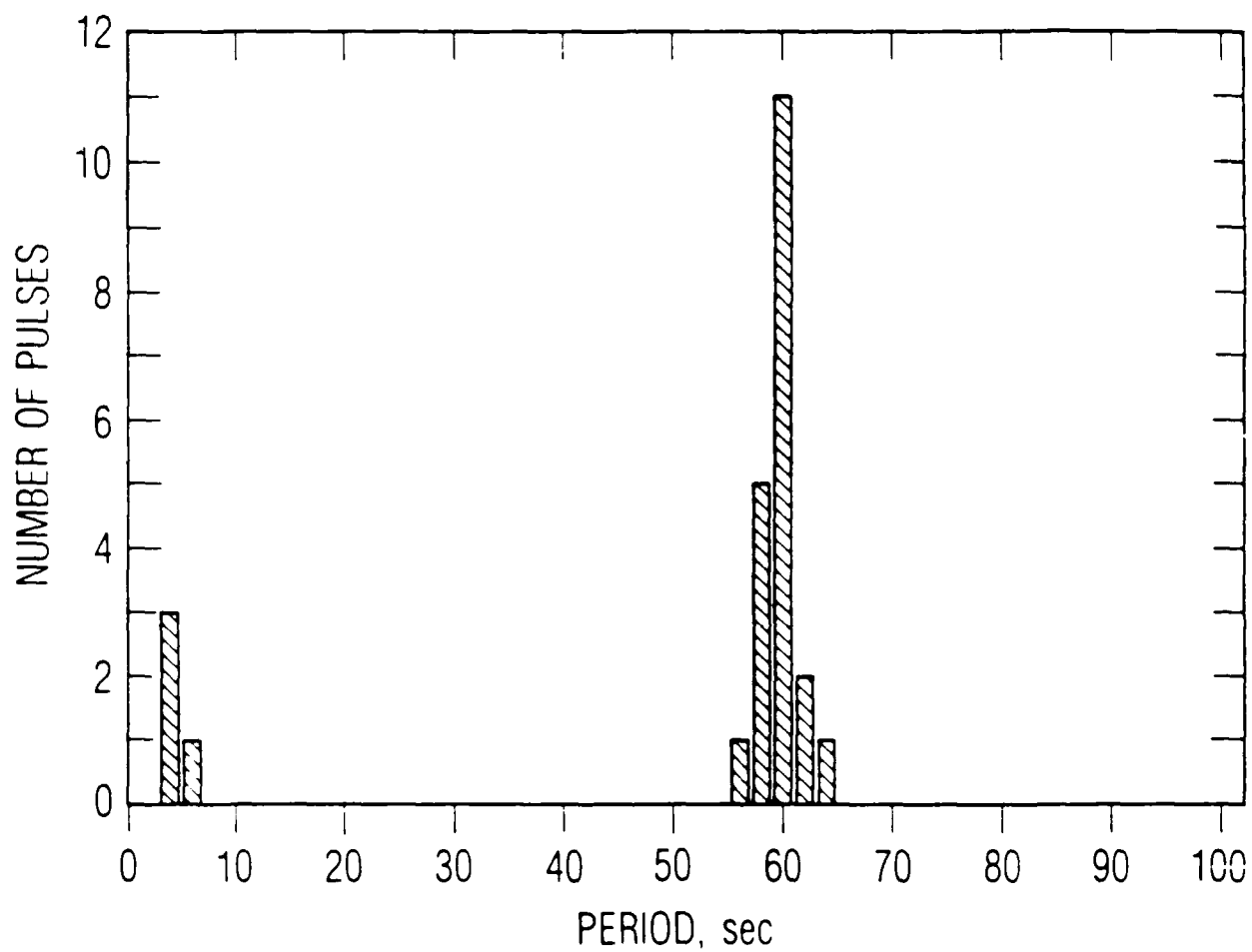


Fig. 7. Histogram of the Repetition Rate of the Discharges Detected by the Pulse Analyzer Between 24059 and 27359 s on 22 September 1982

## VI. THE ENVIRONMENT

### A. DST VALUES

Three large magnetospheric storms occurred during September 1982. Figure 8 shows the hourly equatorial Dst values for the month. The first storm occurred on 6 September. Dst reached a minimum value of  $-303$  gamma at 12 hr UT on 6 September. One pulse was detected by the Pulse Analyzer at 19294 s (05:21) during the onset of the storm on that date. Five more pulses were detected on 7 September during the recovery phase.

The second storm occurred on 22 September. The minimum value for Dst was  $-228$  gamma reached at 08 hr on the 22nd. Although this storm was not as severe as the storm on 6 September, the discharges generated by the interaction with the plasma were much more severe. This was the case even though the satellite was in the same dawn local-time sector during both storms. Small discharges were detected on the 23rd, 24th, and 25th during the recovery phase of this storm. The pulses on the 24th and 25th occurred on the day side of the earth where surface charging has not been observed. These discharges were most likely bulk discharges in a cable bundle.

The third storm occurred on 26 September. The minimum value for Dst was  $-205$  gamma reached at 19 hr on the 26th. The Pulse Analyzer detected only one pulse on that day at 18570 s. This was well before the beginning of the storm.

### B. ELECTRONS

Figure 9 shows the distribution function for electrons from 80 eV to 300 keV on 22 September at 06:44 UT. These measurements were made perpendicular to the magnetic field line during a time period with peak charging levels. A similar curve, a time period with peak charging levels on April 24, 1979 (from Fig. 21a of Ref. 7), is shown for comparison. The charging event on 24 April 1979 was the most severe charging event encountered by the SCATHA satellite prior to 22 September 1982. The distribution functions for the two dates are remarkably similar. However, the distribution function measured on 22 September is significantly higher in the energy range from 1 to 10 keV.

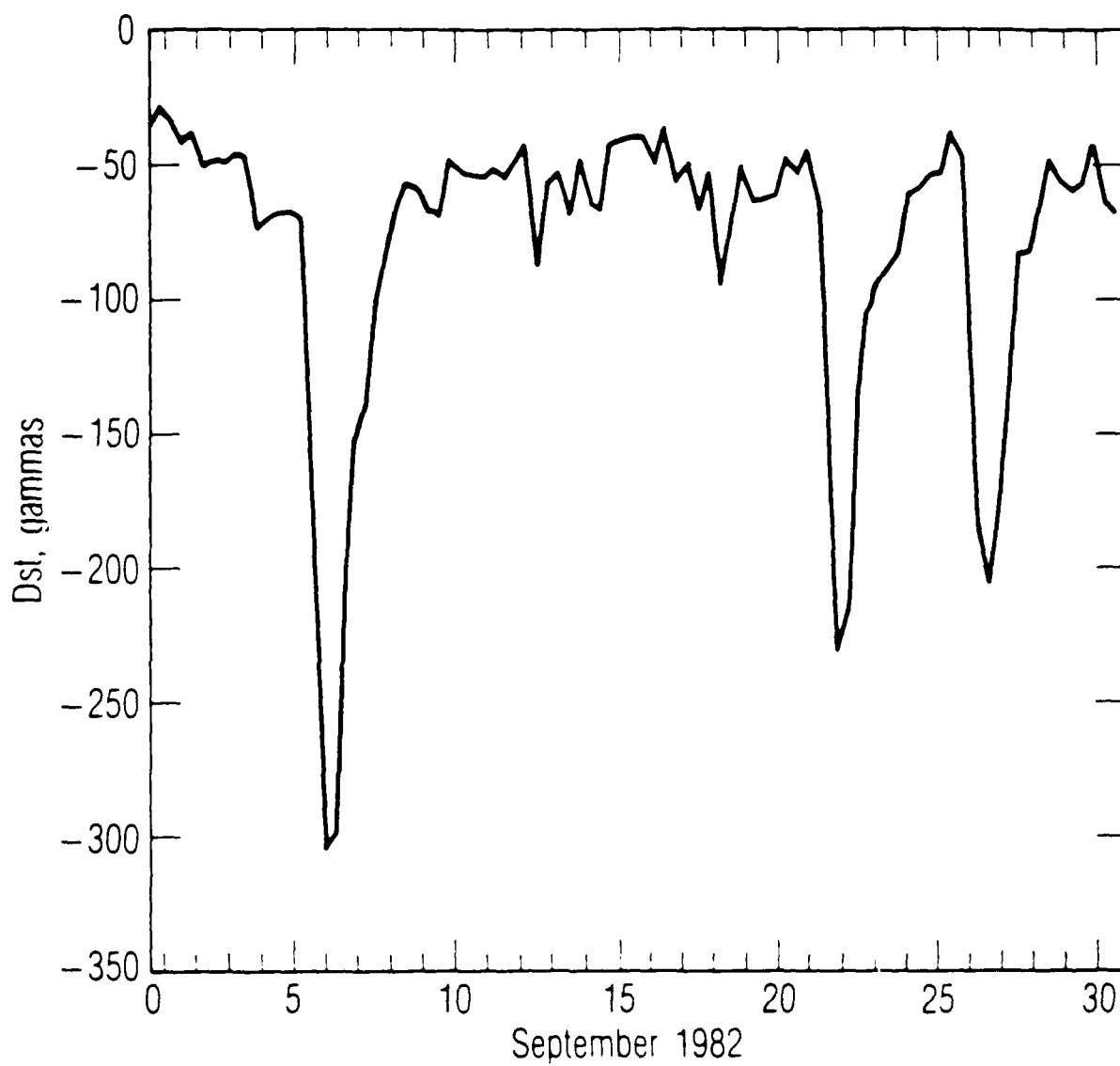


Fig. 8. Hourly Equatorial Dst Values for September, 1982

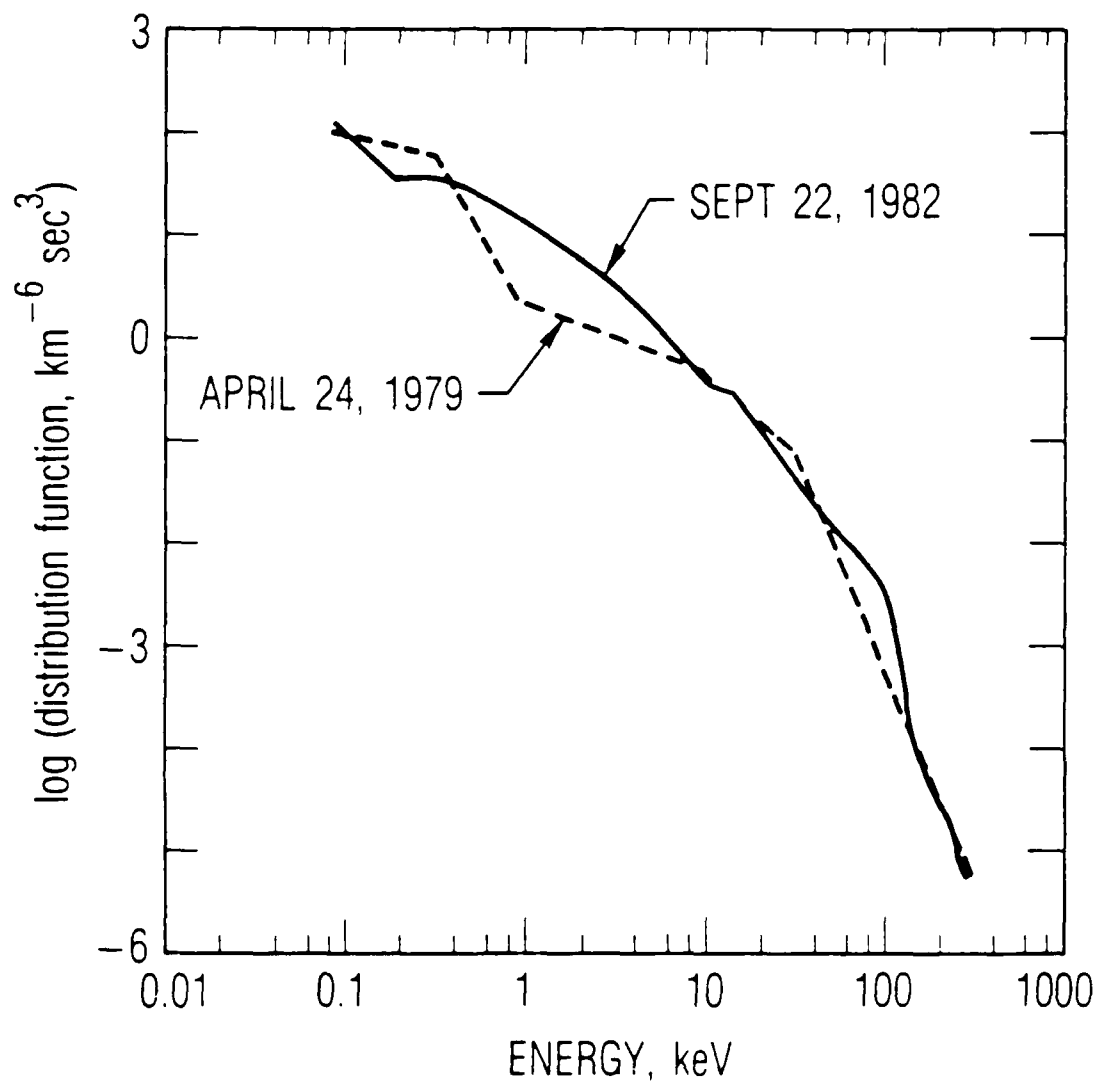


Fig. 9. Distribution Functions of Electrons Measured Perpendicular to the Magnetic Field - (solid) 22 September 1986 at 06:44 UT, (dashed) April 24, 1979 at 06:50 UT

C. MAGNETIC FIELD

Figure 10 shows the magnetic field measured by the Magnetic Field Monitor experiment aboard SCATHA during the time period of interest on 22 September. The measurements show evidence for a large current sheet in the vicinity of the spacecraft just after 07:30 UT.

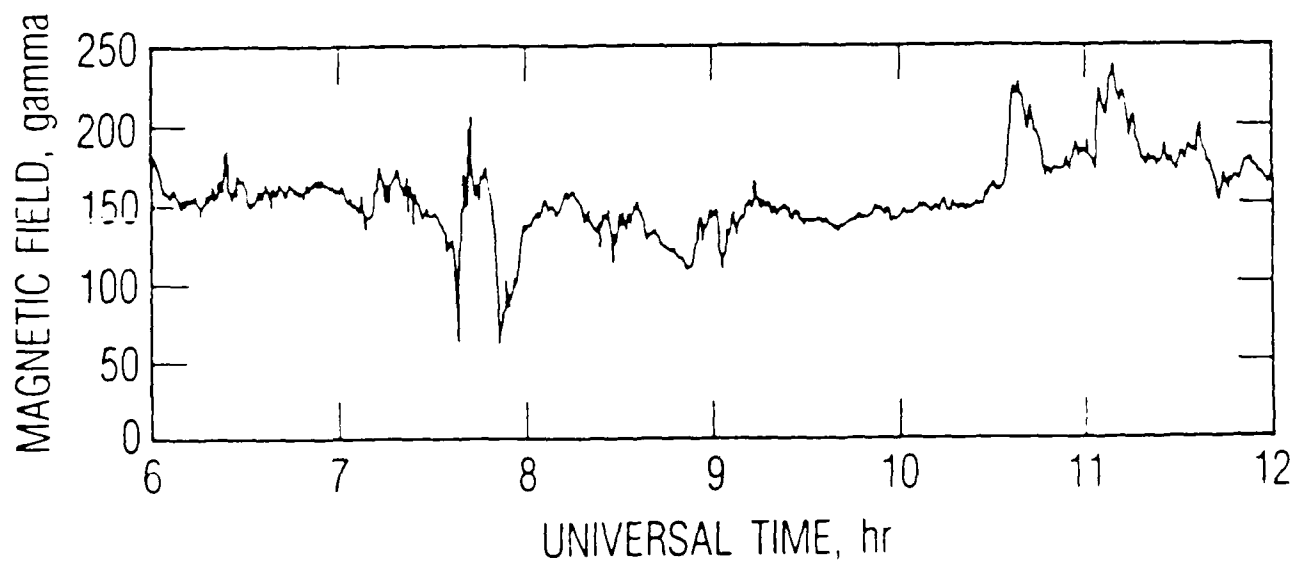


Fig. 10. Magnetic Field at the SCATHA Satellite on 22 September 1982

## VII. SUMMARY

The space plasma environment encountered by the SCATHA satellite on 22 September 1982 caused the highest charging levels and the largest electrostatic discharge pulses detected since the vehicle was launched in 1979. The discharges produced serious anomalies in the operation of some of the scientific instruments and a two-minute loss of data from the vehicle.

The amplitude of the pulses were significantly larger than those measured during the preflight system level tests, which were conducted according to MIL-STD-1541.

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## LABORATORY OPERATIONS

The Aerospace Corporation functions as an "architect-engineer" for national security projects, specializing in advanced military space systems. Providing research support, the corporation's Laboratory Operations conducts experimental and theoretical investigations that focus on the application of scientific and technical advances to such systems. Vital to the success of these investigations is the technical staff's wide-ranging expertise and its ability to stay current with new developments. This expertise is enhanced by a research program aimed at dealing with the many problems associated with rapidly evolving space systems. Contributing their capabilities to the research effort are these individual laboratories:

Aerophysics Laboratory: Launch vehicle and reentry fluid mechanics, heat transfer and flight dynamics; chemical and electric propulsion, propellant chemistry, chemical dynamics, environmental chemistry, trace detection; spacecraft structural mechanics, contamination, thermal and structural control; high temperature thermomechanics, gas kinetics and radiation; cw and pulsed chemical and excimer laser development including chemical kinetics, spectroscopy, optical resonators, beam control, atmospheric propagation, laser effects and countermeasures.

Chemistry and Physics Laboratory: Atmospheric chemical reactions, atmospheric optics, light scattering, state-specific chemical reactions and radiative signatures of missile plumes, sensor out-of-field-of-view rejection, applied laser spectroscopy, laser chemistry, laser optoelectronics, solar cell physics, battery electrochemistry, space vacuum and radiation effects on materials, lubrication and surface phenomena, thermionic emission, photo-sensitive materials and detectors, atomic frequency standards, and environmental chemistry.

Computer Science Laboratory: Program verification, program translation, performance-sensitive system design, distributed architectures for spaceborne computers, fault-tolerant computer systems, artificial intelligence, micro-electronics applications, communication protocols, and computer security.

Electronics Research Laboratory: Microelectronics, solid-state device physics, compound semiconductors, radiation hardening; electro-optics, quantum electronics, solid-state lasers, optical propagation and communications; microwave semiconductor devices, microwave/millimeter wave measurements, diagnostics and radiometry, microwave/millimeter wave thermionic devices; atomic time and frequency standards; antennas, rf systems, electromagnetic propagation phenomena, space communication systems.

Materials Sciences Laboratory: Development of new materials: metals, alloys, ceramics, polymers and their composites, and new forms of carbon; non-destructive evaluation, component failure analysis and reliability; fracture mechanics and stress corrosion; analysis and evaluation of materials at cryogenic and elevated temperatures as well as in space and enemy-induced environments.

Space Sciences Laboratory: Magnetospheric, auroral and cosmic ray physics, wave-particle interactions, magnetospheric plasma waves; atmospheric and ionospheric physics, density and composition of the upper atmosphere, remote sensing using atmospheric radiation; solar physics, infrared astronomy, infrared signature analysis; effects of solar activity, magnetic storms and nuclear explosions on the earth's atmosphere, ionosphere and magnetosphere; effects of electromagnetic and particulate radiations on space systems; space instrumentation.

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